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DyRoBeS© *Manual* | RotorBal

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Influence coefficient method is used in the balancing calculation. The theory is based on two papers:

1. Tessarzik, J. M., Badgley, R. H., and Anderson, W. J., 1972, Flexible Rotor Balancing by the Exact Point-Speed Influence Coefficient Method, ASME Journal of Engineering for Industry, Feb., 1972, pp 148-158.
2. Lund, J. W. and Tonnesen, J., 1972, Analysis and Experiments on Multi-Plane Balancing of a Flexible Rotor, ASME Journal of Engineering for Industry, Feb., 1972, pp 233-242.

Since the least square method is used to solve the simultaneous equations, the Number of Measured Probes times the Number of Speed Points must be greater than or equal to the Number of Balancing Planes. ($N_s X N_m \geq N_b$).

To use the influence coefficient method, no prior knowledge in rotor mode is required. However, trial weights are required to obtain the influence coefficients.

All the inputs are self-explanatory. They are briefly described below:

1. Number of Balancing Planes: Nb

The balancing planes are the planes along the rotor where the trial weights and balancing corrections are applied. Note that the trial weight can be **left-in** or **removed** after the trial run.

2. Number of Measured Probes: Nm

The measurement probes are where the vibrations are taken and recorded. The purpose of the balancing is to find the optimal balancing corrections at the balancing planes such that the vibrations at the measurement probes are minimized.

3. Number of Speeds/Cases: Ns

The number of speeds or cases allows for different speeds or cases, such as idle speed, full speed, full load, unloads, etc. . .

4. Runout Compensation

Runout can be included or excluded in this balancing calculation.

5. Comments

Up to 3 comment lines can be used to describe the system under study.

6. Graphic Data

The following graphic data are used for the graphic purposes which do not affect the calculation results. These data can be changed in the postprocessor if necessary.

6.1. Shaft Rotation: CCW or CW**6.2. Phase: Lag or Lead**

Phase Lag indicates that the phase angle increases against the shaft rotation direction, and Phase Lead indicates that the phase angle increases with the shaft rotation direction.

6.3. 0 degree: Up or Right.

0 degree position defines the reference mark where all the angles (phases) are measured from

6.4. Number of Balancing Holes

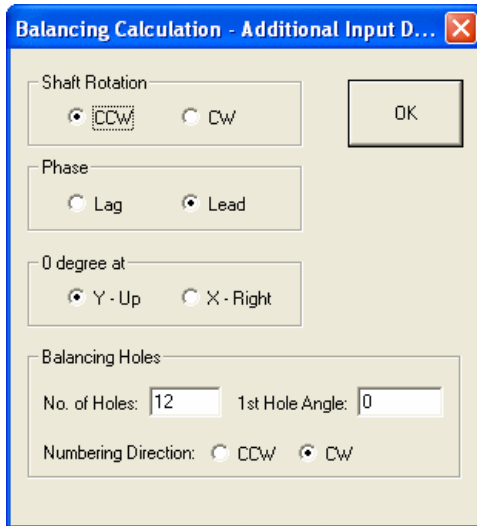
To be used if the balance weight is to be divided into two holes.

6.5 1st Hole Angle

Phase angle of the first hole from the zero degree position. Measured according to Phase Lag or Phase Lead in 6.2.

6.6 Number Direction

The hole numbering direction can be either CCW or CW.



Balancing Calculation - Additional Input D...

Shaft Rotation: CCW CW

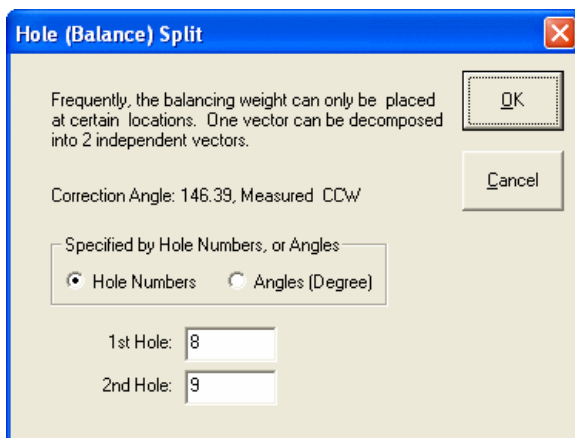
Phase: Lag Lead

0 degree at: Y - Up X - Right

Balancing Holes:
No. of Holes: 1st Hole Angle:

Numbering Direction: CCW CW

OK



Hole (Balance) Split

Frequently, the balancing weight can only be placed at certain locations. One vector can be decomposed into 2 independent vectors.

Correction Angle: 146.39, Measured CCW

Specified by Hole Numbers, or Angles:
 Hole Numbers Angles (Degree)

1st Hole:
2nd Hole:

OK
Cancel

7. Weighting Factors

Weighting factor allows one to strengthen or weaken the data from the measurement probes or speeds. For example, one may use higher weighting factors for the probes where the critical components are located and/or speeds where the rotor will be operated most of the time. Weighting factor zero indicates that the specific probe data will not be included in the calculation.

DyRoBeS-RotorBal: C:\1620\Ch_9_Example_2_TwoPlaneBalancingWithRunout.BAL

Number of Balancing Planes: 2 Number of Speeds/Cases: 1 Shaft Rotation: CCW Phase: Lag 0 degree at: Y-Up X-Right

Number of Measured Probes: 2 Runout Compensation: Yes

Comment: Handbook of Rotordynamics, Example 3.11, pp3.90

Comment: Handbook of Rotordynamics, Example 3.11, pp3.90

Comment: Handbook of Rotordynamics, Example 3.11, pp3.90

Balancing Holes: No. of Holes: 36 1st Hole Angle: 0 Numbering Direction: CCW CW

Weighting (Scale) Factors for probes and speeds

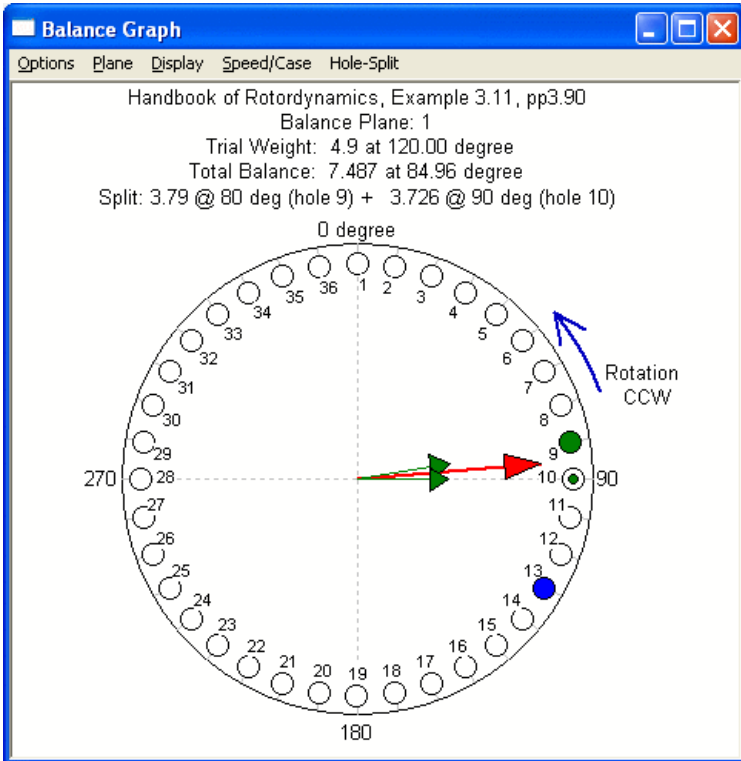
Probe	Factor	Speed	Factor
1	1	1	1
2	1	2	
3		3	
4		4	
5		5	
6		6	
7		7	
8		8	
9		9	
10		10	
11			
12			
13			
14			
15			
16			
17			
18			
19			
20			

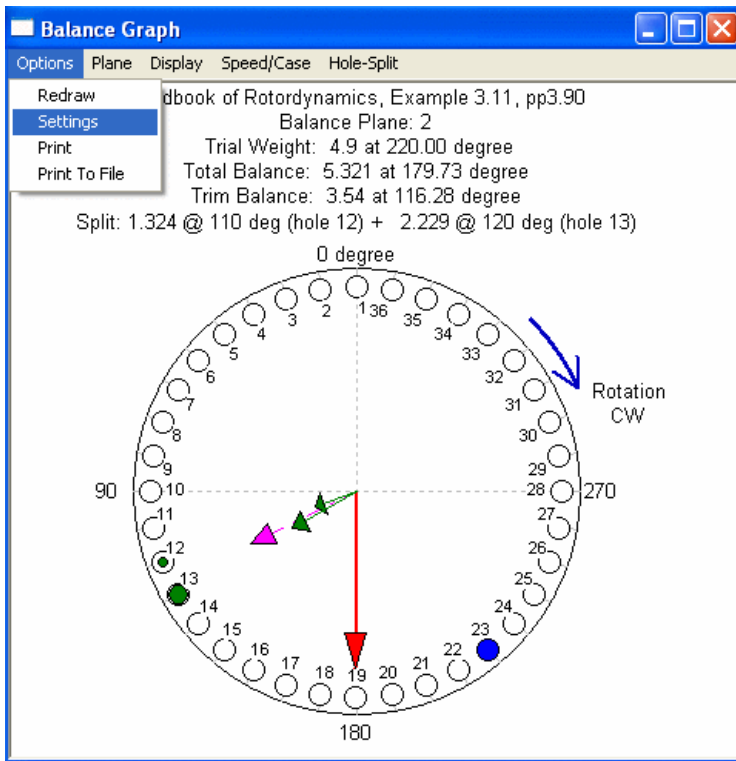
	Condition	Speed	Description	Amplitude	Phase (deg)
1	-- Runout --	---	Probe: 1	0.5	272
2	-- Runout --	---	Probe: 2	0.4	123
3	Initial Readings	1	Probe: 1	1.8	148
4	Initial Readings	1	Probe: 2	3.6	115
5	Trial Run < 1 >	---	Remove Afterward	4.9	120
6	Response	1	Probe: 1	1.1	178
7	Response	1	Probe: 2	2	98
8	Trial Run < 2 >	---	Left-In Afterward	4.9	220
9	Response	1	Probe: 1	2.1	98
10	Response	1	Probe: 2	3.7	102
11					
12					
13					
14					
15					
16					
17					
18					
19					
20					

New Open

Save Save As

Run Close





Balance Graph

Options: Plane Display Speed/Case Hole-Split

Redraw: dbook of Rotordynamics, Example 3.11, pp3.90

Settings: Balance Plane: 2

Print: Trial Weight: 4.9 at 220.00 degree

Print To File: Total Balance: 5.321 at 179.73 degree
Trim Balance: 3.54 at 116.28 degree

Split: 1.324 @ 110 deg (hole 12) + 2.229 @ 120 deg (hole 13)

0 degree

Rotation CW

90 180 270

WithRunout.BAL

Shaft Rotation: CCW CW

Phase: Lag Lead

0 degree at: Y - Up X - Right

Balancing Holes: No. of Holes: 36 1st Hole Angle: 0

Numbering Direction: CCW CW

Weighting (Scale) Factors for probes and speeds

Balancing Calculation - Additional Input D...

Shaft Rotation: CCW CW

Phase: Lag Lead

0 degree at: Y - Up X - Right

Balancing Holes: No. of Holes: 36 1st Hole Angle: 0

Numbering Direction: CCW CW

Factor: 1

OK

Open

Save As

Close

The figure shows the same circular diagram as the first image, but with a settings dialog box open. The dialog box has several sections: Shaft Rotation (CCW selected), Phase (Lag selected), 0 degree at (Y - Up selected), Balancing Holes (No. of Holes: 36, 1st Hole Angle: 0, Numbering Direction: CW selected), and Weighting (Scale) Factors for probes and speeds (Factor: 1). There are also buttons for OK, Open, Save As, and Close.

8. PostProcessor Graphic Data

The following graphic data are used for the graphic purposes which do not affect the calculation results. These data can be changed in the postprocessor if necessary.

8.1. Shaft Rotation: CCW or CW

8.2. Phase: Lag or Lead

Phase Lag indicates that the phase angle increases against the shaft rotation direction, and Phase Lead indicates that the phase angle increases with the shaft rotation direction.

8.3. 0 degree: Up or Right.

0 degree position defines the reference mark where all the angles (phases) are measured from.

8.4. Number of Balancing Holes

To be used if the balance weight is to be divided into two holes. Any vector can be decomposed into two independent vectors.

8.5 1st Hole Angle

Phase angle of the first hole from the zero degree position. Measured according to Phase Lag or Phase Lead in 6.2.

8.6 Number Direction

The hole numbering direction can be either CCW or CW.

Balancing Calculation - Additional Input D... ✕

Shaft Rotation
 CCW CW OK

Phase
 Lag Lead

0 degree at
 Y - Up X - Right

Balancing Holes
No. of Holes: 1st Hole Angle:
Numbering Direction: CCW CW

Hole (Balance) Split ✕

Frequently, the balancing weight can only be placed at certain locations. One vector can be decomposed into 2 independent vectors. OK

Correction Angle: 146.39, Measured CCW Cancel

Specified by Hole Numbers, or Angles
 Hole Numbers Angles (Degree)

1st Hole:
2nd Hole:

Example 1:

DyRoBeS_RotorBal - Input Data

Number of Balancing Planes: Number of Speeds/Cases:

Number of Measured Probes: Runout Compensation:

Shaft Rotation: CCW CW

Phase: Lag Lead

0 degree at: Y - Up X - Right

Weighting (Scale) Factors for probes and speeds

Comment: Handbook of Rotordynamics, Example 3.11, pp3.90

Comment: Runout compensation is included

Comment: The first trial weight is removed afterward, 2nd trial weight is left-in

	Condition	Speed	Description	Amplitude	Phase (deg)
1	-- Runout --	---	Probe: 1	0.5	272
2	-- Runout --	---	Probe: 2	0.4	123
3	Initial Readings	1	Probe: 1	1.8	148
4	Initial Readings	1	Probe: 2	3.6	115
5	Trial Run < 1 >	---	Remove Afterward	4.9	120
6	Response	1	Probe: 1	1.1	178
7	Response	1	Probe: 2	2	98
8	Trial Run < 2 >	---	Left-In Afterward	4.9	220
9	Response	1	Probe: 1	2.1	98
10	Response	1	Probe: 2	3.7	102
11					
12					
13					
14					
15					
16					
17					
18					
19					
20					

Probe	Factor
1	1
2	1
3	
4	
5	
6	
7	
8	
9	
10	
11	
12	
13	
14	
15	
16	
17	
18	
19	
20	

Speed	Factor
1	1
2	1
3	1
4	1
5	1
6	1
7	
8	
9	
10	
11	
12	
13	
14	
15	
16	
17	
18	
19	
20	

Buttons: New, Open, Save, Save As, Run, Close

Balancing Calculation

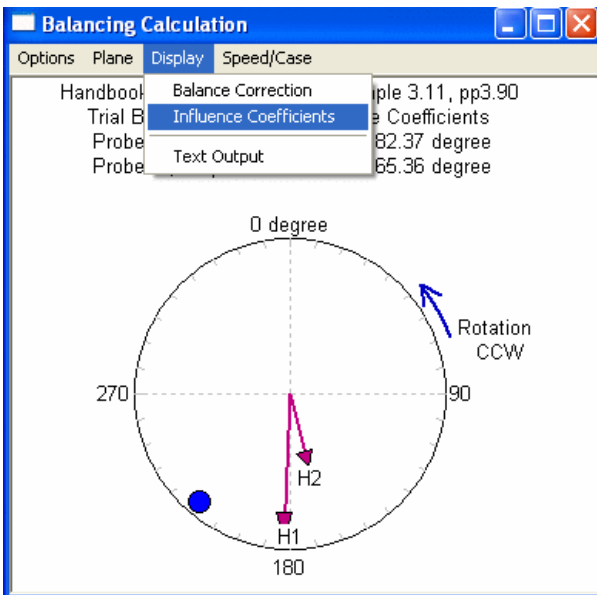
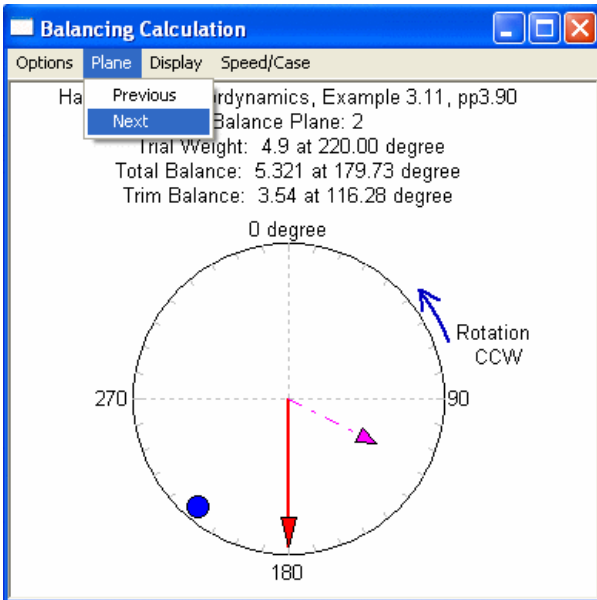
Options Plane Display Speed/Case

Handbook of Rotordynamics, Example 3.11, pp3.90

Balance Plane: 1

Trial Weight: 4.9 at 120.00 degree

Total Balance: 7.487 at 84.96 degree



Handbook of Rotordynamics Example 3.11, pp 3.90

Runout Compensation is included

The first trial weight is removed afterward, 2nd trial weight is left-in

```

**** Number of Speeds or Cases : 1
**** Number of Balancing Planes : 2
**** Number of Measurement Probes: 2

**** Runout (slow-roll vectors) ****
Probe      Amplitude      Phase Angle
1          0.50000      272.00
2          0.40000      123.00

===== Initial Response (Without Trails) =====
Speed  Probe      Amplitude      Phase Angle
1      1          1.8000         148.00
1      2          3.6000         115.00

***** Trial Unbalance Run: 1 *****
Plane  Amplitude      Phase Angle      Afterward
1      4.9000         120.00          Remove

----- Response to Trial Unbalance -----
Speed  Probe      Amplitude      Phase Angle
1      1          1.1000         178.00
1      2          2.0000         98.000

```

***** Trial Unbalance Run: 2 *****

Plane	Amplitude	Phase Angle	Afterward
2	4.9000	220.00	Left-In

----- Response to Trial Unbalance -----

Speed	Probe	Amplitude	Phase Angle
1	1	2.1000	98.000
1	2	3.7000	102.00

*** Weighting Factors for probes and speeds ***

Probe	Weighting Factor
1	1.0000
2	1.0000

Speed	Weighting Factor
1	1.0000

=====
<<<<<<<< Total Balance Correction >>>>>>>>

Correction Required to Balance the Rotor

Plane No.	Amplitude	Phase Angle
1	7.4873	84.957
2	5.3209	179.73

<<<<<<<< Trim Balance Correction >>>>>>>>

Correction Required if Trial Weight Left-in

Trim Balance = Total Balance - Trial Weight

Plane No.	Amplitude	Phase Angle
2	3.5402	116.28

=====
***** The Influence Coefficients *****

Trial-Run	Speed	Probe	Influence Coef.	
			Amplitude	Phase
1	1	1	0.20617	175.
1	1	2	0.36446	194.
2	1	1	0.34092	182.
2	1	2	0.16986	165.

=====
===== Predicted Residual Response =====

Speed	Probe	----- WithOUT Runout -----		----- With Runout -----	
		Amplitude	Phase Angle	Amplitude	Phase Angle
1	1	0.0000	0.0000	0.50000	272.00
1	2	0.0000	0.0000	0.40000	123.00

=====

Example 2:

DyRoBeS_RotorBal - Input Data

Number of Balancing Planes: 2 Number of Speeds/Cases: 6 Shaft Rotation: CCW CW Phase: Lag Lead

Number of Measured Probes: 2 Runout Compensation: No

Comment: Example from ROTORBAL Example 5.3.2 - 70 MW Gas Turbine

Comment: Example from ROTORBAL Example 5.3.2 - 70 MW Gas Turbine

Comment: Example from ROTORBAL Example 5.3.2 - 70 MW Gas Turbine

0 degree at: Y - Up X - Right

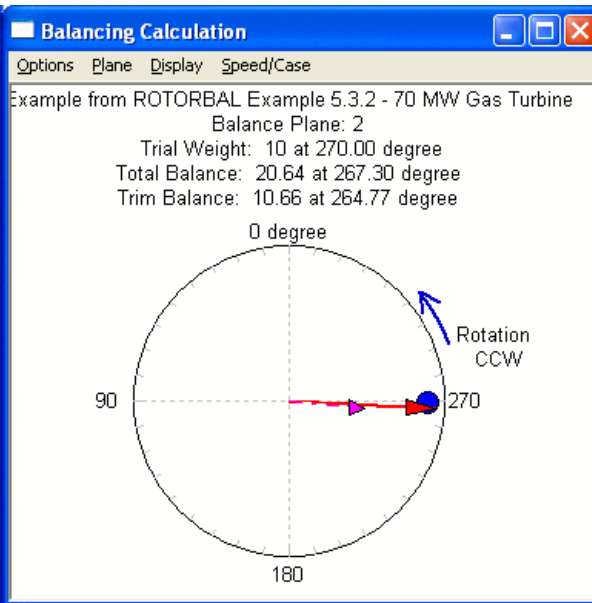
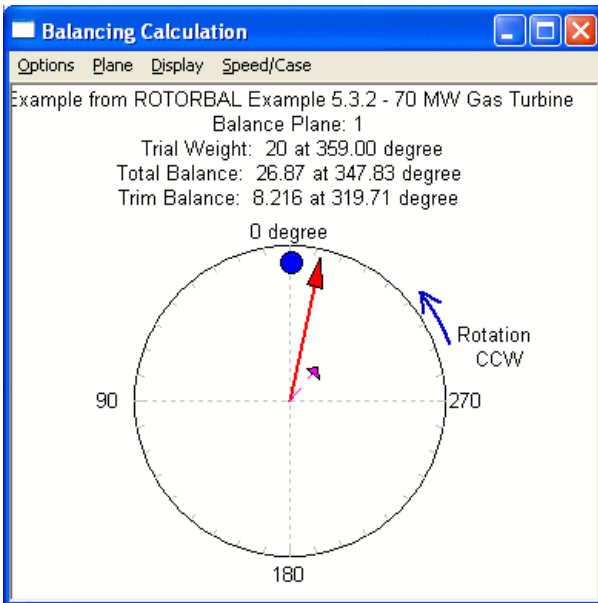
Weighting (Scale) Factors for probes and speeds

Condition	Speed	Description	Amplitude	Phase (deg)	Probe	Factor	Speed	Factor	
1	Initial Readings	1	Probe: 1	1.7	339	1	1	1	1
2	Initial Readings	1	Probe: 2	4.6	54	2	1	2	1
3	Initial Readings	2	Probe: 1	2.8	226	3		3	1
4	Initial Readings	2	Probe: 2	6.7	10	4		4	1
5	Initial Readings	3	Probe: 1	3.9	145	5		5	1
6	Initial Readings	3	Probe: 2	3.7	333	6		6	1
7	Initial Readings	4	Probe: 1	4.5	103	7		7	
8	Initial Readings	4	Probe: 2	4.7	302	8		8	
9	Initial Readings	5	Probe: 1	5.4	74	9		9	
10	Initial Readings	5	Probe: 2	6.5	113	10		10	
11	Initial Readings	6	Probe: 1	1.98	98	11		11	
12	Initial Readings	6	Probe: 2	5.7	114	12		12	
13	Trial Run < 1 >	---	Left-In Afterward	20	359	13		13	
14	Response	1	Probe: 1	2.6	313	14		14	
15	Response	1	Probe: 2	5.9	7	15		15	
16	Response	2	Probe: 1	3.9	232	16		16	
17	Response	2	Probe: 2	4.4	4	17		17	
18	Response	3	Probe: 1	4.2	160	18		18	
19	Response	3	Probe: 2	2.8	340	19		19	
20	Response	4	Probe: 1	3.5	120	20		20	

New Open

Save Save As

Run Close



Example from ROTORBAL Example 5.3.2 - 70 MW Gas Turbine
 2 probes at 6 speeds
 No runout

```
***** Number of Speeds or Cases : 6
***** Number of Balancing Planes : 2
***** Number of Measurement Probes: 2
```

***** NO Runout Compensation

===== Initial Response (Without Trails) =====

Speed	Probe	Amplitude	Phase Angle
1	1	1.7000	339.00
1	2	4.6000	54.000
2	1	2.8000	226.00
2	2	6.7000	10.000
3	1	3.9000	145.00

3	2	3.7000	333.00
4	1	4.5000	103.00
4	2	4.7000	302.00
5	1	5.4000	74.000
5	2	6.5000	113.00
6	1	1.9800	98.000
6	2	5.7000	114.00

***** Trial Unbalance Run: 1 *****

Plane	Amplitude	Phase Angle	Afterward
1	20.000	359.00	Left-In

----- Response to Trial Unbalance -----

Speed	Probe	Amplitude	Phase Angle
1	1	2.6000	313.00
1	2	5.9000	7.0000
2	1	3.9000	232.00
2	2	4.4000	4.0000
3	1	4.2000	160.00
3	2	2.8000	340.00
4	1	3.5000	120.00
4	2	5.4000	325.00
5	1	4.1000	73.000
5	2	3.7000	97.000
6	1	1.5000	141.00
6	2	3.1000	99.000

***** Trial Unbalance Run: 2 *****

Plane	Amplitude	Phase Angle	Afterward
2	10.000	270.00	Left-In

----- Response to Trial Unbalance -----

Speed	Probe	Amplitude	Phase Angle
1	1	1.8000	319.00
1	2	4.3000	15.000
2	1	3.1000	244.00
2	2	3.4000	9.0000
3	1	3.2000	107.00
3	2	1.9000	329.00
4	1	2.4000	122.00
4	2	4.4000	330.00
5	1	2.4000	61.000
5	2	3.5000	101.00
6	1	1.0200	170.00
6	2	3.1100	104.00

*** Weighting Factors for probes and speeds ***

Probe	Weighting Factor
1	1.0000
2	1.0000

Speed	Weighting Factor
1	1.0000
2	1.0000
3	1.0000
4	1.0000
5	1.0000
6	1.0000

=====
 <<<<<<<< Total Balance Correction >>>>>>>>

Correction Required to Balance the Rotor

Plane No.	Amplitude	Phase Angle
1	26.867	347.83
2	20.639	267.30

=====
 <<<<<<<< Trim Balance Correction >>>>>>>>

Correction Required if Trial Weight Left-in

Trim Balance = Total Balance - Trial Weight

Plane No.	Amplitude	Phase Angle
1	8.2159	319.71
2	10.660	264.77

=====

***** The Influence Coefficients *****

Trial-Run	Speed	Probe	Influence Coef.	
			Amplitude	Phase
1	1	1	0.65281E-01	279.
1	1	2	0.21766	317.
1	2	1	0.57655E-01	248.
1	2	2	0.11846	202.
1	3	1	0.54915E-01	228.
1	3	2	0.49103E-01	134.
1	4	1	0.77078E-01	242.
1	4	2	0.10636	26.
1	5	1	0.65130E-01	258.
1	5	2	0.15575	313.
1	6	1	0.67568E-01	230.
1	6	2	0.14110	312.
2	1	1	0.83143E-01	210.
2	1	2	0.17475	257.
2	2	1	0.10809	105.
2	2	2	0.10554	258.
2	3	1	0.34210	118.
2	3	2	0.10027	271.
2	4	1	0.11046	26.
2	4	2	0.10867	214.
2	5	1	0.18221	359.
2	5	2	0.32108E-01	318.
2	6	1	0.78362E-01	12.
2	6	2	0.27106E-01	279.

===== Predicted Residual Response =====

Speed	Probe	----- WithOUT Runout -----		----- With Runout -----	
		Amplitude	Phase Angle	Amplitude	Phase Angle
1	1	1.0968	310.43	1.0968	310.43
1	2	2.6067	358.29	2.6067	358.29
2	1	2.8744	257.03	2.8744	257.03
2	2	1.7963	40.778	1.7963	40.778
3	1	4.8509	66.019	4.8509	66.019
3	2	0.73525	317.09	0.73525	317.09
4	1	1.6051	154.41	1.6051	154.41
4	2	4.3282	341.01	4.3282	341.01
5	1	0.55031	343.11	0.55031	343.11
5	2	2.1098	114.00	2.1098	114.00
6	1	1.6347	207.09	1.6347	207.09
6	2	2.0997	118.93	2.0997	118.93

Example 3:

DyRoBeS_RotorBal - Input Data

Number of Balancing Planes: 2 Number of Speeds/Cases: 2 Shaft Rotation: CCW CW Phase: Lag Lead

Number of Measured Probes: 4 Runout Compensation: No 0 degree at: Y - Up X - Right

Comment: W501 Gas Turbine 108 MW

Comment: 2 Speeds, 3070 RPM, 3600 RPM

Comment: 4 Probes and 2 Balancing Planes

	Condition	Speed	Description	Amplitude	Phase (deg)
1	Initial Readings	1	Probe: 1	1.1	143
2	Initial Readings	1	Probe: 2	3.7	268
3	Initial Readings	1	Probe: 3	1.7	287
4	Initial Readings	1	Probe: 4	0.8	156
5	Initial Readings	2	Probe: 1	3.7	98
6	Initial Readings	2	Probe: 2	7.5	41
7	Initial Readings	2	Probe: 3	3.9	1
8	Initial Readings	2	Probe: 4	4.2	209
9	Trial Run < 1 >	---	Left-In Afterward	19.5	185
10	Response	1	Probe: 1	2.5	70
11	Response	1	Probe: 2	1.9	216
12	Response	1	Probe: 3	2.9	23
13	Response	1	Probe: 4	1.9	216
14	Response	2	Probe: 1	2.5	2
15	Response	2	Probe: 2	6.8	350
16	Response	2	Probe: 3	9.4	359
17	Response	2	Probe: 4	6.3	202
18	Trial Run < 2 >	---	Left-In Afterward	7.4	70
19	Response	1	Probe: 1	2.3	301
20	Response	1	Probe: 2	4.4	216

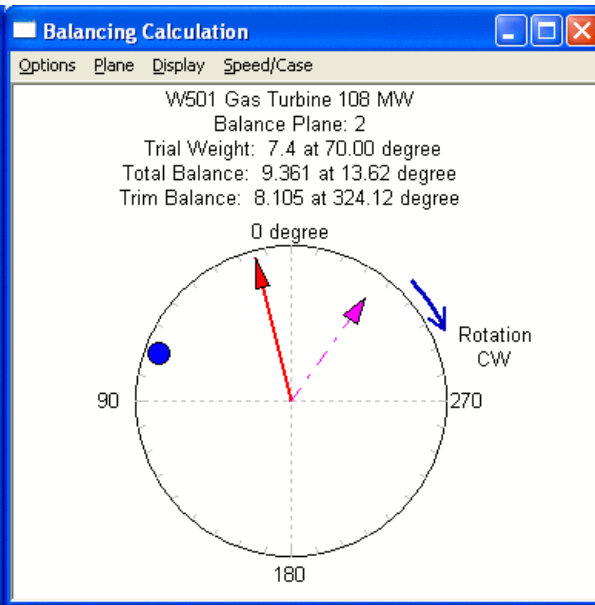
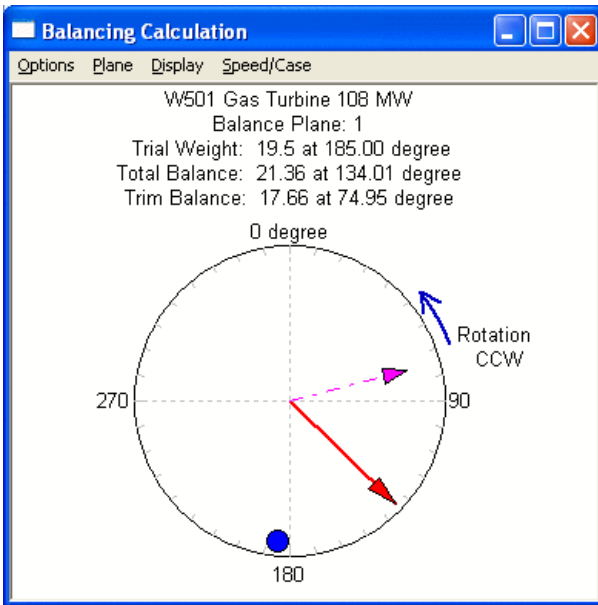
Probe	Factor
1	1
2	1.2
3	1.1
4	1
5	
6	
7	
8	
9	
10	
11	
12	
13	
14	
15	
16	
17	
18	
19	
20	

Speed	Factor
1	1
2	1.2
3	
4	
5	
6	
7	
8	
9	
10	

New Open

Save Save As

Run Close



W501 Gas Turbine 108 MW
2 Speeds, 3070 RPM, 3600 RPM
4 Probes and 2 Balancing Planes

***** Number of Speeds or Cases : 2
***** Number of Balancing Planes : 2
***** Number of Measurement Probes: 4

***** NO Runout Compensation

===== Initial Response (Without Trails) =====

Speed	Probe	Amplitude	Phase Angle
1	1	1.1000	143.00
1	2	3.7000	268.00
1	3	1.7000	287.00
1	4	0.80000	156.00
2	1	3.7000	98.000

2	2	7.5000	41.000
2	3	3.9000	1.0000
2	4	4.2000	209.00

***** Trial Unbalance Run: 1 *****

Plane	Amplitude	Phase Angle	Afterward
1	19.500	185.00	Left-In

----- Response to Trial Unbalance -----

Speed	Probe	Amplitude	Phase Angle
1	1	2.5000	70.000
1	2	1.9000	216.00
1	3	2.9000	23.000
1	4	1.9000	216.00
2	1	2.5000	2.0000
2	2	6.8000	350.00
2	3	9.4000	359.00
2	4	6.3000	202.00

***** Trial Unbalance Run: 2 *****

Plane	Amplitude	Phase Angle	Afterward
2	7.4000	70.000	Left-In

----- Response to Trial Unbalance -----

Speed	Probe	Amplitude	Phase Angle
1	1	2.3000	301.00
1	2	4.4000	216.00
1	3	2.5000	294.00
1	4	0.80000	139.00
2	1	1.7000	12.000
2	2	5.6000	355.00
2	3	5.3000	344.00
2	4	3.8000	181.00

*** Weighting Factors for probes and speeds ***

Probe Weighting Factor

1	1.0000
2	1.2000
3	1.1000
4	1.0000

Speed Weighting Factor

1	1.0000
2	1.2000

=====
<<<<<<<< Total Balance Correction >>>>>>>>

Correction Required to Balance the Rotor

Plane No.	Amplitude	Phase Angle
1	21.356	134.01
2	9.3609	13.616

=====
<<<<<<<< Trim Balance Correction >>>>>>>>

Correction Required if Trial Weight Left-in

Trim Balance = Total Balance - Trial Weight

Plane No.	Amplitude	Phase Angle
1	17.665	74.947
2	8.1048	324.12

=====
***** The Influence Coefficients *****

-- Before application of Weighting Factors --

Influence Coef.

Trial-Run	Speed	Probe	Amplitude	Phase
1	1	1	0.12406	219.
1	1	2	0.15077	294.
1	1	3	0.18008	227.
1	1	4	0.84732E-01	56.
1	2	1	0.23984	125.
1	2	2	0.31737	95.
1	2	3	0.28226	173.
1	2	4	0.11241	3.
2	1	1	0.58558	204.
2	1	2	0.33784	146.

2	1	3	0.51293	174.
2	1	4	0.25519	350.
2	2	1	0.11851	92.
2	2	2	0.17773	78.
2	2	3	0.60743	127.
2	2	4	0.41498	338.

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===== Predicted Residual Response =====
----- WithOUT Runout -----
Speed   Probe   Amplitude   Phase Angle   Amplitude   Phase Angle
1       1       3.9669      49.648       3.9669      49.648
1       2       1.9065      348.46      1.9065      348.46
1       3       2.2628      79.442      2.2628      79.442
1       4       0.22558     305.47      0.22558     305.47
2       1       1.8109      9.5156      1.8109      9.5156
2       2       1.8740      232.93      1.8740      232.93
2       3       3.3169      159.79      3.3169      159.79
2       4       1.8632      30.989      1.8632      30.989
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