# Dyrobes

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# Examples for Ver 19.00

Three examples are used to demonstrate the new features implemented in Ver19.00 and some other commonly used features while performing the rotordynamics analysis.

## Case 1 – Air Compressor

The rotor and bearing used in this example are shown below.





Note that the special characters, such as dot "." and white space "", are allowed in the file and path name. The bearing coefficients are calculated using BePerf from 5,000 to 50,000 rpm with an increment of 5,000 rpm. After the bearing run, the bearing coefficients can be outputted and then imported into the Rotor program.

Number of Pads: 3

Leading Edge: 89

Trailing Edge: 189

<u>N</u>ew

Bearing Data for Pad #1

<u>O</u>pen

Preload: 0.6

Offset: 0.95

<u>S</u>ave

Inlet Temperature: 120

Heat carried away: 80

Advanced Fetaures

No

Save <u>A</u>s

(degF)

<u>C</u>lose

[%]

<u>R</u>un



There is a Check Box named "Reverse direction of rotation" when outputting (saving) the bearing coefficients. Do <u>NOT</u> check this box unless you are performing a coupled Lateral-Torsional vibration analysis on a geared system and this particular shaft is rotating clockwise. Referring the previous bearing figure, all the bearing coefficients are calculated based on the shaft counter-clockwise rotation. For a single rotor system, this is always the case. For more details on the coupled lateral and torsional analysis, see book "Practical Rotordynamics and Fluid Film Bearing Design", Chapter 8.

The bearing coefficients can be imported into the Rotor program using the Bearing Type 2 – Bearing Coefficients From Data File (Rotor – Model – Data Editor – Bearings –

Select Type 2 – then Browse the bearing file outputted from BePerf). In Ver 19.00, the interpolation method for the speed dependent bearing coefficients can be either spline function or linear interpolation. Spline function was used before Ver 19.00.

Rotor Bearing System Data	×
Axial Forces   Static Loads   Constraints   Misalignments   Shaft Bow   Time Forcing Units/Description   Material   Shaft Elements   Disks   Unbalance   Bearings   Supports	Harmonics Torsional/Axial Foundation User's Elements
Bearing: 1 of 2 Foundation Add Brg Del Brg	Previous Next
Station I: 2 J: 0 Angle: 0	
Type: 2-Bearing Coefficients From Data File	it 💌
Comment: Bearing at collar end	
FileName: C:\V1900\mytest.data\example files\Case 1\AirCompressor_Brg1.brg	Browse
Open	2 🛛
Look in: 🗁 Case 1 💌 🖛 🖻	<b>*</b>
AirCompressor_Brg1.brg	
Brg1.brg	
Brg2.brg	
File name: AirCompressor_Brg1.brg	Open
Files of type: Bearing K,C Files (*.brg)	Cancel It: Lbf-s/in
SaveSave As	

To keep the file simple and portable, we use Bearing Type 1- Speed Dependent Bearing and import the bearing coefficients from the BePerf output. If you have the bearing output file from BePerf program, this can be easily done by open the \*.brg (output file from BePerf) using MS Excel, delete the first header line, save the file into xls format, then use the Import \*.xls button in Rotor Bearing Tab to import the bearing coefficients. Note that this option is not commonly selected since you have all the necessary rotor and bearing data files and Bearing Type 2 is more suitable for this purpose. In some cases, if you obtain the bearing coefficients from other sources or if you need to share your rotor file but not the bearing data, then you can enter the bearing coefficients as the speed dependent bearing (Type 1).

Rotor Be	earing Sys	tem Data									X
Axial Fr Units/D	orces   S) Description   ing: 1 of -2	tatic Loads   Material   S	Constraints Shaft Element	Misalignm s   Disks	ents Shaft Unbalance	Bow T Bearings	ime Forcing Support	g Hann ts Foun	nonics   1 dation   U	Forsional/A: Iser's Eleme	kial Ints
Static	and: 2	1: IO		i Foundai	ion <u>t</u>	Add Brg	DeiBr		evious	Export * xls	
T	ype: 1-Spe	ed Dependent	Bearing				- Splin	e Fit	<b>_</b>		1
Comm	nent: Bearin	g at collar end									
	rpm	Kxx	Кхи	Kux	Kuu	Cxx	Схи	Сух	Cyy	Max	
1	5000.00	87216.40	14558.50	-160580.00	285233.00	170.64	-83.73	-83.73	627.34	0	
2	10000.00	128020.00	46582.80	-173625.00	288828.00	153.06	-25.97	-25.97	329.24	0	
3	15000.00	173365.00	79036.50	-197574.00	321930.00	145.85	-12.81	-12.81	252.78	0	
4	20000.00	218035.00	109104.00	-223148.00	360335.00	140.23	-7.74	-7.74	216.76	0	
5	25000.00	260018.00	136497.00	-247835.00	398538.00	134.87	-5.24	-5.24	194.44	0	
6	30000.00	299556.00	161846.00	-271369.00	435515.00	129.98	-3.81	-3.81	178.74	0	
7	35000.00	336837.00	185491.00	-293711.00	470937.00	125.54	-2.92	-2.92	166.79	0	
8	40000.00	372089.00	207687.00	-314925.00	504771.00	121.48	-2.31	-2.31	157.23	0	
9	45000.00	405531.00	228636.00	-335105.00	537093.00	117.76	-1.88	-1.88	149.30	0	
10	50000.00	437360.00	248498.00	-354348.00	568010.00	114.34	-1.57	-1.57	142.56	0	
11											
12											
13											
14											
15											-
•										•	
Inse	ert Row	Delete Row	]					Unit:(2)	) - Kt: Lbf/in	, Ct: Lbf-s/i	n
						Save	Save <u>A</u>	s	<u>C</u> lose	<u>H</u> elp	

The first analysis illustrated here is the Critical Speed Map, which is required by API. Note that the "spin/whirl ratio = 1" in the analysis input indicates the calculated critical speeds are the Synchronous Forward Critical Speeds commonly excited by the mass unbalance. The result can be viewed using the PostProcessor. The operating speed line and the bearing dynamic stiffness in the critical speed map can be plotted using the menu – Options – Settings. The bearing dynamic stiffness including damping effect is recommended in this map and required by the API. There are two intersection points (critical speeds) according to this map; one is around 23,000 rpm caused by  $K_{dx}$  and the other is around 26,000 rpm caused by  $K_{dy}$ . Due to significant damping in this case and the bearings are non-isotropic, there will only one critical speed around 26,000 rpm be observed in the unbalance response curve and machine test. In general, the higher one with more shaft flexibility will be observed.

Shaft Element Effects       Time       Frequency       Domain       Frequency       X:       0         Static Deflection       Critical Speed Map       Constrained Bearing Stations       Critical Speed Map       X:       0         Static Deflection       Critical Speed Map       Time-Start:       0       V       Mass Unbalance         V:       Constrained Bearing Stations       Critical Speed Map       Time-Start:       0       V       Mass Unbalance         Spin/Whirl Ratio:       1       Bearing K · Min:       1000       Increment:       1e+006       V       Mass Unbalance         No. of Modes:       5       Max:       1e+008       Solution Method       Gravity (Z)       None zero         Whirl Speed and Stability Analysis       State Up State Synchronous Response Analysis       Vertical R       Vertical R         Whirl Speed and Stability Analysis       Steady State Synchronous Response Analysis       Steady State Harmonic Excitation         RPM-Starting:       10000       Ending:       40000       V       Mass Unbalance         Increment:       500       Increment:       100       Vertical R       Run	Analysis: 3 - Critical Speed Map	•	🗆 Transient Analysis –		. – Gravity (n) – – –
Static Deflection       Critical Speed Map         Image: Constrained Bearing Stations       Critical Speed Map         Image: Critical Speed Analysis       Spin/Whirl Ratio: 1         Bearing K - Min: 1000       Bearing K - Min: 1000         No. of Modes: 5       Max: 1e+008         Stiffness:       Kxxd dynamic Image: All         Image: Cancer and Stability Analysis       Steady State Synchronous Response Analysis         Whirl Speed and Stability Analysis       Steady State Synchronous Response Analysis         RPM-Starting:       10000         Increment:       1000         Ending:       40000         Increment:       100         Solution Method       Ending: 40000         Increment:       100	Shaft Element Effects	iormation 🔽 Gyroscopic 🔲 Gz	RPM: 35000	Time Frequency Domain	X: 0
Whirl Speed and Stability Analysis     Steady State Synchronous Response Analysis     Steady State Harmonic Excitation       RPM-Starting:     10000     Effects:     RPM-Starting:     10000       Ending:     40000     Image: Analysis     Image: Analysis     RPM-Starting:     10000       Increment:     500     Increment:     100     Image: Analysis     RPM-Starting:     10000       Increment:     500     Increment:     100     Image: Analysis     Run	Static Deflection Constrained Bearing Stations Critical Speed Analysis Spin/Whirl Ratio: 1 No. of Modes: 5 Brg Stiffness: Kxxd dynamic @ rpm: 23000	Critical Speed Map Spin/Whirl Ratio: 1 Bearing K - Min: 1000 Npts: 0 Max: 1e+008 Stiffness to be varied at Bearings: All Allow Bearings in Series	Time-Start: 0 Ending: 0.05 Increment: 1e-006 Solution Met Newmark-be Initial Cs: No	An Speed, Societ phil Mass Unbalance Const. Unbalance Shaft Bow Disk Skew Gravity (X,Y) hod Gravity (Z) sta Time Forcing Misalignment	Y: -386.088 Z: 0 None zero Gz Vertical Rotor
No. of Modes: 6 Excitation Shaft: 1 I Disk Skew Excitation Shaft: 1 Cancel	Whirl Speed and Stability Analysis         RPM-Starting:       10000         Ending:       40000         Increment:       500         No. of Modes:       6	Steady State Synchronous Response RPM-Starting: 10000 Ending: 40000 Increment: 100 Excitation Shaft: 1 V	e Analysis Effects: Mass Unbalance Const. Unbalance Shaft Bow Disk Skew Micalianment	Steady State Harmonic Excitation RPM-Starting: 10000 Ending: 40000 Increment: 100 Excitation Shaft: 1	n Run Cancel



Critical Speed Map	
Title: Critical Speed Map	OK ]
X-Label: Bearing Stiffness	Cancel
Y-Label: Critical Speeds (rpm)	
Modes: 1.2.3	For example: 1,2,3,5,7,9 or 3
For Selected Modes, enter mode number	rs separated by commans.
Operating Speed Range: 40000	to O
🔽 Print Spin/Whirl Ratio	
🔽 Curve Fit 🔲 Symbol 🔽 Maj	ior Grid 🛛 🗖 Minor Grid
🥅 Manual Scaling	
Manual Scaling Data	
Xmin: 1000 Ymin: 1000	
Xmax: 1e+008 Ymax: 1e+00	06 Uverlay Brg Stiffness
XDiv: 5 YDiv: 5	

Be	aring	Stiffness									×
	<ul> <li>Impor</li> <li>D</li> <li>Brows</li> <li>Brows</li> </ul>	t from the exi irect Stiffnes e C:\V e C:\V	isting DyRoBe s - K 1900\mytest.c 1900\mytest.c	S Brg files Dynamic Stil Jata\example fi Jata\example fi	fness - (K^2+ les\Case 1\4 les\Case 1\4	·(Cw)^2 )^0.5 .irCompressor_ .irCompressor_	Brg1.brg Brg2.brg	Reset	Export Data	ОК	
L	Co	lor Number:	40	40		40	40		40	40	
		rpm-1	Kxx-1	Куу-1	rpm-2	Kxx-2	Куу-2	rpm-3	Kxx-3	Күү-3	-
	1	5000	124859	435031	5000	130387	456826	0	0	0	
	2	10000	205136	449768	10000	207937	463477	0	0	0	
	3	15000	287306	511170	15000	288731	521913	0	0	0	
	4	20000	365775	579603	20000	366399	588808	0	0	0	
	5	25000	438488	646492	25000	438590	654763	0	0	0	
	6	30000	506447	710620	30000	506174	718254	0	0	0	
	7	35000	570240	771689	35000	569684	778857	0	0	0	
	8	40000	630380	829787	40000	629601	836605	0	0	0	
	9	45000	687313	885132	45000	686355	891667	0	0	0	
	10	50000	741421	937962	50000	740316	944269	0	0	0	
	11										
	12			0							
	13			0	0						
	14									• • • • • • • • • • • • • • • • • • • •	
	15										-

To examine the critical speed mode shapes, the critical speed analysis is performed. First, let us examine the critical speed around 26,000 rpm estimated from the Critical Speed Map. In Ver 19.00, the **dynamic** stiffness option is added and the speed can be specified. Before Ver 19, only the **direct** stiffness and the bearing coefficients from the Last speed point (for the speed dependent coefficients) were used. For compatible with the old versions, the speed ZERO can be entered to indicate the last speed point will be used. From the energy distribution, the bearing at impeller end has much more potential energy than that of the collar end (33.7% vs. 5.2%). Therefore, this bearing is more critical in the design. For the critical speed around 23,000 rpm (in the x-direction), enter 23.000 rpm and select *Kxx*dynamic in the stiffness option. Most of the potential energy in this case is in the bearings and the shaft only has 36% potential energy. This mode is more close to the rigid bode mode compared with the mode at the 26,000 rpm, therefore, this mode is not observed in the unbalance response curve, which will be shown later.

In Ver 19, the bearing coefficients after interpolation and used in the analysis are also printed in the text output for reference. A warning message will appear if the analysis speed is outside the speed range of the bearing coefficients. Again, in the Critical Speed Analysis, the ZERO speed indicates the last speed point will be used as before ver 19.

Brg Coefficients after Interpolation No. rpm Kxx Kxy 1 26000. 268110. 141715. 
 Kxy
 Kyx
 Kyy
 Cxx
 Cxy
 Cyx
 Cyy

 141715.
 -252637.
 406053.
 133.853
 -4.91026
 -4.91026
 190.982
 2 26000. 269498. 137247. -259252. 410936. 133.480 -6.44488 -6.44488 193.375 \_\_\_\_\_ \* Critical Speed Analysis \* \*\*\* Y-Direction Properties are specified \*\*\* Dynamic Stiffness = sqrt(Kyy<sup>2</sup>+(omegaCyy)<sup>2</sup>) Bearing Coefficients Used @ Speed = 26000.00 RPM Stiffness Kt = 659747.46 Kr = 0.0000000Bearing No: 1 Bearing No: 2 Stiffness Kt = 667890.00 Kr = 0.0000000\* \* \* \* \* \* \* Spin(1)/Whirl Ratio = 1.000 \* \* \* \* \* \* \* R/S rpm Ηz no 2765.53 6628.20 10295.3 31625.3 26408.9 1 440.148 2 63294.6 1054.91 3 98313.0 1638.55 4 302000. 5033.33 61798.5 9835.53 5 590132. 0.101755E+07 106558. 6 16959.2 7 0.176134E+07 184447. 29355.7 57149.7 8 0.342898E+07 359082. 9 0.463345E+07 485214. 77224.2 

Analysis: 2 - Critical Speed Analysis	Transient Analysis	- Gravity (g)
Shaft Element Effects  Rotatory Inertia  Shear Deformation  Gyroscopic  G	RPM: 35000 Time Domain Domain	×: 0
Static Deflection     Critical Speed Map       Constrained Bearing Stations     Spin/Whirl Ratio:       Critical Speed Analysis     Bearing K - Min:       Spin/Whirl Ratio:     1       Npts:     0       May:     1e+008	Time-Start:       0       ✓       Mass Unbalance         Ending:       0.05       ✓       Shaft Bow         Increment:       1e-006       ✓       Disk Skew	Y: -386.088 Z: 0 None zero Gz
No. of Modes:     5     Stiffness to be varied at       Brg Stiffness:     Kyyd dynamic ▼     Kyyd dynamic ▼       @ rpm:     26000     Kxxx Kyy       Whirl Speed and Stability Analysis     Kxxd dynamic	Solution Method Gravity (Z) Newmark-beta Static Loads Initial Cs: No Misalignment Steady State Harmonic Excitation	
Kyoo gonamic         Kyoo gonamic           RPM-Starting:         10000         [Kxxd+Kyyd]/2];         10000           Ending:         40000         Ending:         40000           Increment:         500         Increment:         100	Effects:     RPM-Starting:     10000       ✓     Mass Unbalance     Ending:     40000       ✓     Const. Unbalance     Increment:     100       ✓     Shaft Bow     Increment:     100	Run
No. of Modes: 6 Excitation Shaft: 1 All Synchronized Shafts	Image: Disk Skew         Excitation Shaft:         1           Image: Misalignment         Image: All Shafts with same speed	Cancel
Steady Maneuvers (Base Constant Translational Acceleration and/or Ti Speed (RPM): 35000 Acceleration - X: 0 Y: 773	m Rate) .176 Tum Rate - X: 0 Y: 0 Ref Pos:	0









The next analysis is the unbalance response analysis. The analysis speed starts from 10,000 rpm to 40,000 rpm with an increment of 100 rpm. Since this speed increment (100 rpm) is smaller than the speed increment (5,000 rpm) in the bearing coefficients, interpolation will be implemented in the program. The interpolation method (spline or linear) is specified in the Bearing input tab. The vibration at the probe station (station 5) is shown in the Bode plot (PostProcessor – Steady Synchronous Response – Bode Plot). Some graphic settings can be adjusted under Options – Settings. In this example, English units are used. The response displacement is in inches. But, it is common to use mils in US to describe the vibration amplitude. Therefore, we entered 1000 in the Amplitude Scale to convert the inches to mils. The Text Color box is checked to show the corresponding color in the text printout in the plot. The curve color can also be changed in the Preference Settings – Post-Processor Graph Colors. Also, the amplitude printout is formatted with floating point with 2 decimal points. Some of the graphic settings can be pre-defined and saved in the Preference Settings File (Project - Preference Settings -Post-Processor Graph Settings and Post-Processor Graph Colors). These graphic features are implemented in all the post-processor plots when applicable. Again, use Options -Settings to make necessary adjustments to meet your needs.

Lateral Analysis Option & Run of Analysis: 5 - Steady State Synchron Shaft Element Effects Rotatory Inertia Shear Def Static Deflection Constrained Bearing Stations Critical Speed Analysis Spin/Whitl Ratio: 1 No. of Modes: 5 Brg Stiffness: Kyyd dynamic @ rpm: 26000	Time Data          ous Response - Linear System         iormation         Image: Critical Speed Map         Spin/Whirl Ratio:         Image: Spin/Whirl Ratio:         Image: New York         Image: New York     <	Transient Analysis RPM: 35000 Const Time-Start: 0 Ending: 0.05 Increment: 1e-006 Solution Mel Newmark-bu Initial Cs: No	Time Domain Domain ant Speed: 35000 rpm Mass Unbalance Const. Const. Con	Gravity (g) X: 0 Y: -386.088 Z: 0 None zero Gz Vertical Rotor
Whirl Speed and Stability Analysis RPM-Starting: 10000 Ending: 40000 Increment: 500	Steady State Synchronous Respons RPM-Starting: 10000 Ending: 40000 Increment: 100	e Analysis Effects: Mass Unbalance Const. Unbalance Shaft Bow	- Steady State Harmonic Excitation RPM-Starting: 10000 Ending: 40000 Increment: 100	Run
No. of Modes:  P Steady Maneuvers (Base Constant T Speed (RPM): 35000 Accel	Excitation Shaft:     I       Image: All Synchronized Shafts     Image: Shafts       ranslational Acceleration and/or Turn R       eration -X:     0       Y:     772.176	Disk Skew Misalignment ate) Turn Rate - X: 0	All Shafts with same speed      Y: 0 Ref F	Cancel



🔲 Bode Plot	Bode Plot (Amplitude and Phase)
Options Station	
Redraw	Title: Bode Plot
Settings	X-Label: Rotational Speed (rpm) Cancel
Print Print to File	Y-Label: Amplitudes (mils)
Export Data	Z.J. abat: Phaces (Deg)
List Peaks	
9	Station: 5 Sub: 1 Relative to Station: 0 Sub: 0
ຍິ 180 - ຊິ	Probe Angles (deg) - X (1): 0 Probe Y (2): 90
<u> </u>	Operating Speed Range: 40000 to 0
050	Safe Margin Speeds: 0 0
	Label Amplification Factor: 🔲 🗙 (1) 🔽 Y (2) 🔽 Mark AF lines
E U.40	Phase Display: 💿 Academic 🕜 Instrument (ADRE)
୍ଟ୍ର 0.30 ଅଟି	✓ Peak - Peak Phase Shift - X(1): 0 Y(2): 0
.20	🗖 Curve Fit 🔲 Symbol 🔽 Major Grid 📄 Minor Grid 🔽 Text Color
₹ 0.10	Manual X Scale: 1 Amplitude Scale: 1000
0.00 5	Manual Scaling Data
	Xmin:         5000         Ymin:         0         Zmin:         0
	Xmax: 45000 Ymax: 0.5 Zmax: 360
	XDiv: 4 YDiv: 5 ZDiv: 4
	Decimal Places for the Response Graphic Printout
	Amplitude: 2 - F:decimals  Phase: 0 - Integer







#### Case 2 – Turbine – Generator Set

This example is a turbine-generator set. The model was built in three separate rotors (files) as shown below. It is noticed that the individual Rotor 1 was modeled in the reversed direction (right to left) compared to the entire train. This can occur frequently since all the drawings (data or model) were provided from different suppliers, or modeled by different engineers. In this example, the Model Flipping and Model Combining features will be demonstrated.

![](_page_14_Figure_2.jpeg)

#### Rotor 1

![](_page_14_Figure_4.jpeg)

#### Rotor 2

![](_page_15_Figure_1.jpeg)

## Rotor 3

![](_page_15_Figure_3.jpeg)

**Step 1** is to flip the rotor 1 model before combining with others: Open the original rotor 1 file, Select Model Flipping under Project menu, then Save the flipped model in a different file name (in case, the original file is needed for other purposes). After this is done, the main window will be updated with this new file name and new model.

![](_page_16_Figure_1.jpeg)

Save As		? 🔀
Save in:	🗀 Case 2	▼ ← 🗈 💣 Ⅲ-
Case 2	_ Generator.rot _ Turbine 1.rot _ Turbine 2.rot	
File <u>n</u> ame:	Case 2 _ Turbine 1_Flip.rot	Save
Save as <u>t</u> y	pe: Rotor Files (*.rot)	Cancel

![](_page_17_Figure_0.jpeg)

**Step 2** to combine (merge, append) the Rotor 2 file into this rotor 1: Select Model Combining under Project menu while the Rotor 1 is still an active project in the screen. A dialog box will appear once the Model Combining is selected. Since the rotor 1 is still active and is used as the base model. Click Add Model to add the second rotor and also click Combined Model to save the combined model into a file. Since all three rotors are in-line in this case, we can simply append the rotor 2 into the rotor 1 in a single shaft without generating a second shaft in the model. We can also use Rigid Link option to keep the rotor 2 in a second shaft and rigidly link to Rotor 1. Both will yield the same analysis results. The Merge Material Number box is checked, that is, if the material properties (Rho, E, G, and description) in rotor 2 can be found in the Rotor 1 material library, then the existing material number in Rotor 1 will be used for Rotor 2, and it will not duplicate the same material in another material number. If the Merge Material box is not checked, the material number in rotor 2 will append to the material numbers in Rotor 1 as a new material. After this step, the main window will be updated with this new combined model.

![](_page_18_Figure_0.jpeg)

![](_page_18_Figure_1.jpeg)

<u>Step 3</u> is to combine the generator (rotor 3). Again, Select Model Combining under Project menu and enter the necessary information as shown below to add the generator

into the system. After these two steps, the separate three rotors are combined into a single model.

Combine (merge,	add) Models	
<u>B</u> ase Model	C:\V1900\mytest.data\example files\Case 2\Case 2_ Tunbines 12.rot	
Add Model	C:\V1900\mytest.data\example files\Case 2\Case 2_ Generator.rot	
Combined Model	C:\V1900\mytest.data\example files\Case 2\Case 2 _ Tunbines_Generator.rot	
This feature w Note: The sha The relationsh The relationsh Rigid Li This rer	ill combine the base model and additional model into a single final model. i.e., Base + Add = Combined Model aft speed ratio, axial distance, and Y distance must be manually adjusted to fit your needs. ip between the Last shaft of the Base model and the First shaft of the added model d the first shaft of the added model to the last shaft of the base model into a SINGLE shaft ink - the first shaft of the added model will be rigidly linked to the last shaft of the base model nains two separate shaft entries, although they are Rigidly linked.	ОК
Merge Mat the materai	erial Numbers - if the material properties (rho, E, G) of the Added model and the Base model are the same, I number of the Based model will be used. Otherwise, new material number will be created.	Cancel

![](_page_19_Figure_2.jpeg)

If you prefer to keep these three rotors in separate shafts in the system, then you may use the Rigid Link option instead of the Append in the check box as shown below. Both methods yield the same analysis results.

<u>B</u> ase Model	C:\V1900\mytest.data\example files\Case 2\Case 2_ Tunbines 12_RigidLink.rot	_
Add Model	C:\V1900\mytest.data\example files\Case 2\Case 2 _ Generator.rot	
Combined Model	C:\V1900\mytest.data\example files\Case 2\Case 2_ Tunbines_Generator_RigidLink.rot	
This feature w	di service de la service del cod additerrateradol interación de Costanadol de Deservicadol Continend Madel	
Note: The sha	in combine the base model and additional model into a single linal model. I.e., base + Add = Combined Model ift speed ratio, axial distance, and Y distance must be manually adjusted to fit your needs. ip between the Last shaft of the Base model and the First shaft of the added model	
Note: The sha	III combine the base model and additional model into a single lina model. I.e., base + Add = Combined Model aft speed ratio, axial distance, and Y distance must be manually adjusted to fit your needs. ip between the Last shaft of the Base model and the First shaft of the added model I the first shaft of the added model to the last shaft of the base model into a SINGLE shaft	
Note: The relationsh	In complete the base model and additional model into a single linea model. I.e., base + Add = Completed Model aft speed ratio, axial distance, and Y distance must be manually adjusted to fit your needs. Ip between the Last shaft of the Base model and the First shaft of the added model I the first shaft of the added model to the last shaft of the base model into a SINGLE shaft nk - the first shaft of the added model will be rigidly linked to the last shaft of the base model nains two separate shaft entries, although they are Rigidly linked.	ĸ

![](_page_20_Figure_1.jpeg)

![](_page_21_Figure_0.jpeg)

![](_page_22_Figure_0.jpeg)

![](_page_22_Figure_1.jpeg)

### Case 3 – Rotary Compressor Train

In this example, we will demonstrate the coupled lateral and torsional analysis and the effect of the gear mesh coupling on the system natural frequencies. This system is a motor, which drives a rotatory compressor with a male rotor and a female rotor.

![](_page_23_Figure_2.jpeg)

Again, the rotors are modeled separately as shown below. Each rotor is analyzed first before assembly into a complete system. For each individual rotor, the **lateral** whirl speed analysis is performed to examine the system natural frequencies. They show that all the natural frequencies are far above the rotor operating speed and all potential excitation frequencies. Also, the gyroscopic effect is insignificant in this example. The natural frequencies do not vary much with the rotor speed for all three rotors. However, the coupled lateral and torsional natural frequencies of the entire system (3 rotors together) decreased significantly in this example show below.

### <u>Motor</u>

![](_page_23_Figure_5.jpeg)

🛃 DyRoBeS_Rotor - C:\V1900\mytest. data	\example files\Case 3\Case_3_Motor.rot	
Project Model Analysis PostProcessor Tools V	Lateral Analysis Option & Run Time Data	
Cateral Wibration Constant Axial Wibration Axial Wibration Lateral-Torsional-Axial Wibration	Analysis: 4 - Whirl Speed & Stability Analysis Shaft Element Effects RPM: 0 Rotatory Inettia IV Shear Deformation IV Gyroscopic IV Gz	Time Domain
	Static Deflection       Critical Speed Map         Constrained Bearing Stations       Spin/Whirl Ratio: 1         Critical Speed Analysis       Bearing K - Min: 100         Spin/Whirl Ratio: 1       Npts: 50         No. of Modes: 3       Stiffness to be varied at         Brg Stiffness: [Kxx+Kyy]/2 ]       Allow Bearings in Series	Ant Speed, Orphin
	Whirl Speed and Stability Analysis         RPM-Starting:       0         Ending:       5000         Increment:       500         No. of Modes:       20         Steady Maneuvers (Base Constant Translational Acceleration and/or Turn Rate)         Speed (RPM):       0         Acceleration - X:       0         Y:       0         Turn Rate - X:       0	Steady State Harm RPM-Starting: Ending: Increment: Excitation Shaft: All Shafts with

![](_page_24_Figure_1.jpeg)

![](_page_25_Figure_0.jpeg)

## <u>Compressor – Male Motor</u>

![](_page_26_Figure_1.jpeg)

![](_page_26_Figure_2.jpeg)

![](_page_27_Figure_0.jpeg)

## <u>Compressor – Female Rotor</u>

![](_page_28_Figure_1.jpeg)

![](_page_28_Figure_2.jpeg)

![](_page_29_Figure_0.jpeg)

![](_page_29_Figure_1.jpeg)

## **Combined Gear Train**

Now, let us combine three rotors together. The steps are described below:

1. Open the Motor Model as a base model

![](_page_30_Figure_3.jpeg)

2. Select Model Combining under Project menu and enter the data blow. In this example, three rotors are coupled by the gear meshes. Therefore, do <u>NOT</u> check the Append and Rigid Link boxes. After this is done, the combined model will be shown in the main window. Since the Append and Rigid Link boxes are not checked, the second rotor will be shown in the right of the first rotor with a small gap in between.

Combine (merge, add) Models	×
Base Model C:\V1900\mytest.data\example files\Case 3\Case_3_Motor.rot	
Add Model C:\V1900\mytest.data\example files\Case 3\Case_3_Rotor_1.rot	
Combined Model C:\V1900\mytest.data\example files\Case 3\Case_3_Motor+Rotor1.rot	
This feature will combine the base model and additional model into a single final model. i.e., Base + Add = Combined Model Note: The shaft speed ratio, axial distance, and Y distance must be manually adjusted to fit your needs. The relationship between the Last shaft of the Base model and the First shaft of the added model	
Rigid Link - the first shaft of the added model will be rigidly linked to the last shaft of the base model       OK         This remains two separate shaft entries, although they are Rigidly linked.       OK	
Merge Material Numbers - if the material properties (rho, E, G) of the Added model and the Base model are the same, the material number of the Based model will be used. Otherwise, new material number will be created.	

![](_page_31_Figure_0.jpeg)

3. Since this is a geared system, we need to enter the speed ratio and also adjust the rotor position for proper display. Model – Data Editor – Shaft Elements – Second Shaft. Enter speed ratio, Axial and Y distances.

xial Fo nits/D	orces Iescrip	Sta tion	atic Lo Mater	ads   rial	Constraints Shaft Elements	Misalignment   Disks   U	s   Shaft B Inbalance	ow   Time Bearings   S	Forcing   Ha Supports   For	armonics   Torsional/Ax undation   User's Elemer
Shaft	: 2 of	2		Start	ing Station #:	13 🗆 F	R Link	Add Shaft	Del Shaft	Previous Next
Spee	d Ratio	o: [-1.34	17	A	kial Distance:	757.529	Y Distance:	125		Import *.xls Export *.xls
Com	nent:	Compr	essor	Male R	otor			,		
1	Ela	Sub	Mat	1.69	Length	Mass ID	Mass OD	SHEED	SPEEDD	Comments
1	13	1	1	0	3		39	0	39	Comments
2	13	2	1	ñ	17	n N	43.5	n N	435	
3	13	2	5	1	17	43.5	102 133	43.5	102 133	
4	14	1	1	O	17	0	43.5	0	43.5	
5	14	1	5	1	17	43.5	102.133	43.5	102.133	·····
6	14	2	1	Ō	11	0	43.5	0	43.5	·····
7	15	-1	1	0	21.2	0	43.5	0	55	
8	15	2	1	0	12.65	0	55	0	55	•
9	16	1	1	0	12.65	0	55	0	55	
10	16	2	1	0	19.5	0	69.288	0	69.288	
11	17	1	1	0	34.25	0	116.2	0	98	
12	18	1	1	0	34.25	0	116.2	0	98	
13	19	1	1	0	34.25	0	116.2	0	98	
14	20	1	1	0	34.25	0	116.2	0	98	
15	21	1	1	0	34.25	0	116.2	0	98	
16	22	1	1	0	34.25	0	116.2	0	98	
17	23	1	1	0	34.25	0	116.2	0	98	
18	24	1	1	0	34.25	0	116.2	0	98	
19	25	1	1	0	30.21	0	59.288	0	59.288	
20	25	2	1	0	10.25	0	35.015	0	35.015	<u> </u>
Inser	rt Row	D	elete F	low	ReNumber	Copy & Paste			Unit	:(4) - Length, Diameter: mr

![](_page_32_Figure_0.jpeg)

4. Combine the last rotor into the system. Again, we need to enter the speed ratio and the axial and Y distances in the Shaft Elements tab. We also need to enter the gear mesh data to couple these three rotors.

Combine (merge, add) Models	$\mathbf{X}$
Base Model C:\V1900\mytest.data\example files\Case 3\Case_3_Motor+Rotor1.rot	
Add Model C:\V1900\mytest.data\example files\Case 3\Case_3_Rotor_2.rot	
Combined Model C:\V1900\mytest.data\example files\Case 3\Case_3_Motor+Rotor1+Rotor2.rot	
This feature will combine the base model and additional model into a single final model. i.e., Base + Add = Combined M	lodel
Note: The shaft speed ratio, axial distance, and Y distance must be manually adjusted to fit your needs.	
The relationship between the Last shaft of the Base model and the First shaft of the added model	
Append the first shaft of the added model to the last shaft of the base model into a SINGLE shaft	
🔲 Rigid Link - the first shaft of the added model will be rigidly linked to the last shaft of the base model	ок
This remains two separate shaft entries, although they are Rigidly linked.	
Merge Material Numbers - if the material properties (rho, E, G) of the Added model and the Base model are the sam the material number of the Based model will be used. Otherwise, new material number will be created.	e, Cancel

![](_page_33_Figure_0.jpeg)

Ro	Rotor Bearing System Data											
E	Axial Fo	orces	St	atic Lo	ads	Constraints	Misalignmen	ts   Shaft E	Bow │ Time	Forcing H	armonics	Torsional/Axial
Ľ	Jnits/D	escrip	tion	Mater	rial	Shart Elements	Disks   L	Jinbalance	Bearings   3	upports   Fo	oundation	User's Elements
	Shaft	: 3 of	3		Start	ting Station #: [	29 🗆	R Link	Add Shaft	Del Shaft	Previous	Next
	Speed Ratio:         0.898         Axial Distance:         832.529         Y Distance:         20         Import *.xls         Export *.xls											
	Comn	nent:	Compr	essor	- Fema	le Rotor						
		Ele	Sub	Mat	Lev	Length	MassID	Mass 0D	Stiff ID	Stiff OD	Com	ments
	1	29	1	1	0	3.6	6.8	30.012	6.8	30.012	0011	
	2	29	2	1	Ō	1.505	6.8	28.4	6.8	28.4		
	3	29	3	1	0	8.1975	6.8	30.012	6.8	30.012		
	4	30	1	1	0	8.1975	6.8	30.012	6.8	30.012		
	5	30	2	1	0	8.75	0	43.001	0	43.001		
	6	30	3	1	0	8.75	0	43.001	0	43.001		
	7	31	1	1	0	34.25	0	104	0	95		
	8	32	1	1	0	34.25	0	104	0	95		
	9	33	1	1	0	34.25	0	104	0	95		
	10	34	1	1	0	34.25	0	104	0	95		
	11	35	1	1	0	34.25	0	104	0	95		
	12	36	1	1	0	34.25	0	104	0	95		
	13	37	1	1	0	34.25	0	104	0	95		
	14	38	1	1	0	34.25	0	104	0	95		
	15	39	1	1	0	15.105	0	59.288	0	59.288		
	16	39	2	1	0	15.105	0	59.288	0	59.288		
	17	39	3	1	0	10.6075	0	40.015	0	40.015		
	18	40	1	1	0	8.9375	0	40.015	0	40.015		
	19	40	2	1	0	9.7725	0	40.015	0	40.015	brg	
	20	40	3	1	0	9.7725	0	40.015	0	40.015		-
	Inser	rt Row	D	elete F	Row	ReNumber	Copy & Paste			Un	it:(4) - Lengtł	n, Diameter: mm
	<u>S</u> ave Save <u>A</u> s <u>C</u> lose <u>H</u> elp										<u>C</u> lose	<u>H</u> elp

![](_page_34_Figure_0.jpeg)

Lateral-Torsional-Axial Gear Mesh Couplin	g 🛛 🔀
Gear Mesh: 1 of 2	Add Delete Previous Next <u>OK</u>
Station I: 11 Pitch Diameter: 137.57	© Driving © Driven Gear Pressure Angle: 20
Gear 2 Station J: 14 Pitch Diameter: 102.133	Helix Angle: 29.49
Gear Mesh Stiffness and Da Stiffness	Imping Matrices in (rta) coordinates Thrust Collar (Rider Ring) Damping
K r' ť a'	C r' t' a' Diameter: U
r 0 0 0	r' 0 0 0 Axial K: 0
ℓ 0 388865 O	<u>ν</u> 0 100 0
a' 0 0 0	a' 0 0 0 Axial C: 0
	Unit:(4) - Angle: deg., Length: mm, K:N/mm, C:N-s/mm

Lateral-Torsional-Axial Gear Mesh Coupling	
Gear Mesh: 2 of 2	Add Delete Previous Next DK
Station I: 21 Pitch Diameter: 66.67	Gear Data Pressure Angle: 0
Gear 2 Station J: 35 Pitch Diameter: 100	Helix Angle: 0
Gear Mesh Stiffness and Damp Stiffness	ing Matrices in (r't'a') coordinates Damping
K r' t' a'	C r' t' a' Diameter: 0
r' 0 0 0	r' 0 0 0 Axial K: 0
τ 0 1E+08 0	ε 0 100 0
a' 0 0 0	a' 0 0 0 Axial C: 0
	Unit:(4) - Angle: deg., Length: mm, K:N/mm, C:N-s/mm

![](_page_35_Figure_1.jpeg)

5. The regular bearing with a spring display can complicate the plot. We can display the bearing with a representing small box by changing the bearing length parameter under Project – Preference Settings – Model Display Settings – Bearing Length – make it a NEGATIVE value. Also, noted that one parameter is added in Ver 19 is the L/D ratio. If the element L/D ratio is less than this pre-defined value, a warning message will be prompted.

![](_page_36_Figure_0.jpeg)

![](_page_36_Figure_1.jpeg)

To perform the coupled lateral and torsional analysis, Select Lateral-Torsional-Axial Vibration under the Analysis menu. A dialog box shows up for the analysis inputs. Select Whirl Speed & Stability Analysis. In the Vibration Included group box, check the Lateral and Torsional boxes to include the lateral and torsional vibrations. That is, coupled lateral and torsional vibration through the gear meshes. Check the Convert

Cross-Coupled Bearing Coefficients box. This indicates that if the shaft rotates clockwise, then the bearing coefficients for this shaft will be converted to comply with this direction of rotation. For more details on this conversion, see book "Practical Rotordynamics and Fluid Film Bearing Design", Chapter 8. For a multiple rotor system, the speed input in the analysis dialog is the first shaft speed. The first shaft in DyRoBeS is always the reference shaft. In this example, it is the motor speed. The compressor male rotor is designed with the speed range of 1200-2900 rpm. Therefore, the motor speed range is 891-2153 rpm. The excitations considered are 1X Motor, 1X Compressor (1.347X Motor), and 4X Compressor (5.388X Motor). In this example, there is one interference point at 9660 cpm present within the operating speed range. The X-Axis (speed axis) in the whirl speed map can be converted into the compressor male rotor speed by multiplying the X-Axis by the speed ratio of 1.347 in the Options-Settings. Once the X-axis is converted into the compressor male rotor speed, the 1X Motor excitation is 0.7424 X Compressor excitation. To avoid confusion, a small label "LTA" is shown in the right bottom of the plot to indicate this plot is from the coupled vibration option. The mode shape for this interference point at 9660 cpm frequency is also shown. For the coupled vibration, since the displacements are normalized with different units, different scale may be required. In this example, the torsional displacement is enlarged fifty times. Again, go to Options – Settings to make necessary adjustments.

🛃 DyRoBeS_Rotor - C:\V1900\mytest.data\	example files\Case 3\Case_3_Motor+Rotor1+Rotor2.rot	
Project Model Analysis PostProcessor Tools Vi	ew Help	
Lateral Vibration Torsional Vibration Axial Vibration Lateral-Torsional-Axial Vibration		
Lateral-Torsional-Axial Geared Vibra	tion Run Time Data	X
Analysis: 1 - Whirl Speed & Stability Anal Vibrations Included	Axial Convert Cross-Coupled Bearing Coefficients	ancel <u>B</u> un
- Whirl Speed and Stability Analysis RPM-Starting: 0 Ending: 5000 Increment: 500 No. of Modes: 20	teady State Synchronous Response Analysis IPM-Starting: 6000 Mass Unbalance Ending: 12000 Const. Unbalance Increment: 10 Disk Skew Increment: 10 Disk Skew Increment: 10 Excitation Shaft: 1 All Synchronized Shafts ▼ T/A SS Excitation	Excitation Effects: Lateral Harmonics Torsional Harmonics me speed
Transient A RP Time-St Endir	Analysis (Constant Speed) M: 0 Gravity (X,Y) art: 0 Gravity (Z) art: 0 Gravity (Z) Gravity (Z) Disk Skew Torsional Excitations	
	ent: 🕛 👘 Static Loads 👘 Steady State Excitation	

![](_page_38_Figure_0.jpeg)

Whirl Speed Map 🛛 🗙
Title: Whirl Speed Map OK
X-Label: Motor Rotational Speed (rpm)
Y-Label: Damped Natural Frequencies (rpm)
Number of Modes: 10
Excitation Slopes: 1, 1.347, 5.388 (nX) - For example: 0.5, 1, 2, 15
Operating Speed Range: 891 to 2153
Safe Margin Speeds: 0 0
🔽 Curve Fit 🔲 Symbol 🔽 Major Grid 🔲 Minor Grid
✓ Manual Scaling
Manual Scaling Data X Scale
Xmin:         0         1
Xmax:         3000         Ymax:         30000         Y Scale
XDiv: 5 YDiv: 6 1

![](_page_38_Figure_2.jpeg)

![](_page_38_Figure_3.jpeg)

![](_page_39_Figure_0.jpeg)

From the mode shape shown above, it indicates that this mode is a coupled lateral and torsional mode with very large torsional motion at the compressor and significant lateral motion at the motor driver end. Consequently, torsional excitation from the compressor rotor can cause high lateral vibration in the motor gear end, mainly in the Y direction (line of action) and small vibration in the X direction. With the natural frequency of 9660 cpm, the strong 4X response occurs at the compressor rotor speed of 9660/4=2,415 rpm and motor speed of 9660/5.388=1793 rpm.

To further study this system, in the Lateral-Torsional-Axial vibration run time input, we can check the Torsional vibration only. This will only analyze the torsional vibration with the flexibility in the gear meshes.

Lateral-Torsional-Axial Geared Vibration Run Time Data			×
Analysis: 1 - Whirl Speed & Stability Analysis	<u>C</u> ancel	<u>R</u> un	
Lateral     If Rotation is CW: from Local x to Global X			

![](_page_40_Figure_0.jpeg)

In summary, only the coupled lateral-torsional vibration analysis produces the interference point (potential resonance point) in the whirl speed map (Campbell Diagram). All the individual analysis on the lateral vibration and purely torsional analysis do not have interference point. Since the gyroscopic effect is very small and the frequencies stay almost the same as the speed increases, the following table summarizes the first several frequencies at zero rotor speed for various analyses:

Frequency (cpm) Comparison @ zero rotational speed							
Mode	Torsional (Rigid Link)	Torsional (Flexible Link)	Lateral Motion (Motor)	Lateral Motion (Male Rotor)	Lateral Motion (Female Rotor)	Coupled Vibration	
1			12,905 (X)			9,660 (T,Y)	
2			13,933 (Y) 🗧			12,978 (X,T)	
3	18,857 (T)	17,549 (T)			~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	• 14,537 (Y,T,X)	
4				26,749 (X)		26,749 (X)	
5			27,794 (X)			27,816 (X)	
6			27,942 (Y) 🯼			33,580 (X)	
7					33,580 (X)	34,255 (T,Y)	
8			35,134 (X)			35,128 (X)	
9			38,438 (Y)			38,347 (Y)	
10				47,137 (X)		39,201 (Y,X,T)	
11				47,247 (Y)		48,171 (X,Y,T)	
12					49,434 (Y)	48,278 (Y,T)	
13	72,146 (T)	72,031 (T)			62,232 (X)	62,232 (X)	
14			73,776 (X)			69,440 (X,Y,T)	
15			76,421 (Y)			73,775 (X,T)	

Frequency (cpm) Comparison @ zero rotational speed

Note: T indicates Torsional motion, X lateral motion mainly in X direction, Y lateral motion mainly in Y direction.

In view of the above table and the corresponding mode shapes, the frequency of the motor lateral motion mode (13,933 cpm) in the Y direction (line of action) decreases significantly due to the flexibility introduced by the gear mesh to become the lowest natural frequency (9,660 cpm) of the complete gear train. The frequency in the X direction (12,905 cpm) of the motor lateral motion is nearly unaffected by this torsional coupling effect and becomes 12,978 cpm in the coupled system.

To further study the torsional excitation effect on the rotor lateral vibration, the torsional excitation caused by the compressor is added into the model. The compressor has a 4/6 lobes design for the male and female rotors. It is known that it produces a strong 4X (male rotor speed) port passing torsional excitation from the compressor. This excitation is entered from Model – Data Editor – Torsional/Axial – Steady State Excitation as shown below. Note that the excitations are applied at shaft 2 (compressor male rotor), the excitation frequency is 4X of the compressor male rotor speed, not the motor speed. Also, the exciting torque increases with the square of the compressor speed.

![](_page_42_Figure_0.jpeg)

Torsio	Torsional/Axial Steady State Excitation									
Exci	itation Freq.(c que/Force Mu	pm): Iltiplier:	wo: 0 Ao: 0 ,	w1: 4	w2: 0 <u>D</u> K					
	cpm = w0 + w1 x rpm + w2 x rpm <sup>2</sup> A = A0 + A1 x rpm + A2 x rpm <sup>2</sup> Steady State Harmonic Excitation: T = A * ( Tc cos (wexcT) + Ts sin (wexcT) ) Excitation frequency wexc (rad/sec) = cpm * (2*pi/60). and A is the Amplitude multiplier rpm = excitation shaft speed, rotor speed where the excitation applied									
	T/A	Stn	Cos Component	Sin Component	Comments	*				
1	Torsional	19	1	0	4 X Male Rotor Speed					
2	Torsional	20	1	0						
3	Torsional	21	1	0						
4	Torsional	22	1	0						
5	Torsional	23	1	0						
6										
7				0						
8				0						
9										
10				0						
11										
12										
<u>I</u> nse	12									

When performing the coupled lateral-torsional analysis, select Lateral-Torsional-Axial Vibration from the Analysis menu and enter the data below. Note that the analysis rotor speed is always referred to the shaft 1 rotor speed. Since the excitation is on shaft 2 (compressor male rotor), enter 2 in the excitation shaft field. Check the Torsional Harmonics to include the torsional harmonic excitation in the analysis. Several postprocessor plots are shown below. Although the excitation is due to torsional torque, the lateral vibration can be significant in this coupled gear system.

£	DyRoBeS_Rotor - C:\V1900\mytest.data\example files\Case 3\Case_3_Motor+Rotor1+Rotor2.rot	
Proj	ject Model Analysis PostProcessor Tools View Help	
	Lateral Vibration Torsional Vibration Axial Vibration	
	Lateral-Torsional-Axial Vibration	
	Lateral-Torsional-Axial Geared Vibration Run Time Data	
	Analysis:       4 - Steady State Harmonic Response	
	For a multiple shaft (rotor) system, Shaft one (1) is the reference shaft and the Analysis Speed below is referred to the Shaft 1 Speed.         Whirl Speed and Stability Analysis       Steady State Synchronous Response Analysis         RPM-Starting:       0         Ending:       5000         Increment:       500         Increment:       10         Disk Skew       Excitation Shaft:         Excitation Shaft:       1         Misalignment       Torsional         All Synchronized Shafts       T /A SS Excitation	
	Transient Analysis (Constant Speed)       RPM:       0       Time-Start:       0       Shaft Bow       Ending:       0       Disk Skew       Torsional Excitations	

![](_page_44_Figure_0.jpeg)

Again, by default, the rotor speed shown in the plot is the analysis speed (shaft 1 speed). To plot the vibration vs. the compressor male rotor speed, we can multiply the X-axis by 1.347.

![](_page_44_Figure_2.jpeg)

![](_page_45_Figure_0.jpeg)

![](_page_45_Figure_1.jpeg)

![](_page_46_Figure_0.jpeg)

#### **Carbon Ring Seal Calculation**

The feature calculate the leakage for a carbon ring seal commonly used in the compressors and blowers. The calculation method is based on the air force report by Bauer et al. (1965). The original equations were adjusted with some correction factors to fit the experimental data tested by AMS Seals Inc.

Bauer, P., Glickman, M., and Iwatsuki, F., May 1965, "Analytical techniques for the Design of Seals for Use in Rocket Propulsion Systems, Volume II Dynamic Seals," Technical Report AFRPL-TR-65-61, Air Force Rocket Propulsion Laboratory, Research and Technology Division.

To activate this feature, Rotor – Tools – Estimate the Carbon Ring Seal Leakage.

🛃 DyRoBeS_Rotor - Untitled		
Project Model Analysis PostProcessor	Tools View Help	
	Alford Aerodynamics Wachel Aerodynamics	
	Deep-Groove or Angular Contact Ball Bearing Self-Aligning Ball Bearing Spherical Roller Bearing Straight Roller or Tapered Roller Bearing	
	Liquid Annular Seal - Dynamic Coefficients	
	Estimate Carbon Ring Seal Leakage Activate Laby Seal Calculation: LabySeal Activate Spiral Groove Face Seal: SpiralGF	
	Squeeze Film Damper Design Tools Activate Bearing Program: BePerf Activate Thrust Bearing Program: ThrustBrg	
	Inertia Properties of Homogeneous Solid	
	Equivalent Torsional Stiffness from station I to J	
	Effective Impedance for (Brg+Support) Systems	
	Elliptical Orbit Analysis Total Orbit Motion	
	Balancing Calculation Auto-Balancing Analysis Residual Unbalance per API and ISO	
	- <u> </u>	5
		NUM

Carbon Ring Seal	Leakage Cal	culation				
Comment: DyRoBe	∋S - Rotor					
Units:	English	•	Gas Properties Gas Type: 1 - Air			
Total Seal Width: Shaft Diameter:	0.422	in in	Temperature: 225 degF			
Seal Diameter:	1.2532	in	Viscosity: 3.22629E-09 Lbf-s/in^2	2 B		
High Pressure:	140	psia psia	Eccentricity Ratio : 0.2			
Speed (rpm):	45000	psia	<u>R</u> un <u>C</u> lose			
Results:						
Radial Clearance (C):         0.00165         2C/D × 1000:         2.64021						
Vol. Flov	low Model w: 5.26282	SCFM	Vol. Flow: 5.01289 SCFM			
Mass Flow	M: 0.40198	Lbm/min	Mass Flow: 0.382889 Lbm/min			

#### The Maximum Allowable Residual Unbalance per API and ISO Specifications

The feature calculates the amount of maximum allowable residual unbalance in the rotor per API and ISO specifications. For more details, see book "Practical Rotordynamics and Fluid Film Bearing Design", page 400-401.

To activate the function, Rotor – Tools – Residual Unbalance per API and ISO.

🛃 DyRoBeS_Rotor - Untitled		
Project Model Analysis PostProcessor	Tools View Help	
<u>                                      </u>	Alford Aerodynamics Wachel Aerodynamics	
	Deep-Groove or Angular Contact Ball Bearing Self-Aligning Ball Bearing Spherical Roller Bearing Straight Roller or Tapered Roller Bearing	
	Liquid Annular Seal - Dynamic Coefficients Estimate Carbon Ring Seal Leakage Activate Laby Seal Calculation: LabySeal Activate Spiral Groove Face Seal: SpiralGF	
	Squeeze Film Damper Design Tools Activate Bearing Program: BePerf Activate Thrust Bearing Program: ThrustBrg	
	Inertia Properties of Homogeneous Solid	
	Equivalent Torsional Stiffness from station I to J	
	Effective Impedance for (Brg+Support) Systems	
	Elliptical Orbit Analysis Total Orbit Motion	
	Balancing Calculation Auto-Balancing Analysis Residual Unbalance per API and ISO	
		NUM
The Maximum Allowabl	e Residual Unbalance	

The Maximum Attowable Residual Onbalance				
Speed (rpm): 1 Rotor Weight per Plane: 9	10000 308 LI	• •		<u>C</u> lose
ISO Quality Grade (G-):	2.5 (	G-0.4, 1, 2.5, 6	3, 16, 40, 100, 250	
Umax per ISO		_ Uma	x per API	
G 6.015W/	'n		4W/N	
1.36547 oz-in			0.3632 oz-in	
983.258 gram	n-mm		261.536 gram-r	nm